Assessment of effective thermal conductivity model of the SPACE for fuel axial relocation in ballooned fuel rods

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Background and Objective

- Under LOCA conditions, clad ballooning can occur due to internal overpressure and fuel rods can be overheated and undergo a complex process known as fuel fragmentation, relocation, and dispersal (FFRD) dependent upon fuel burnup.
- The developed model for FFRD phenomena have been

Thermal conductivity Model

- Effective thermal conductivity of crumbled fuel
 - Chiew and Glandt model (implemented in FRAPTRAN)

$$\frac{\lambda_{eff}}{\lambda_{f}} = \frac{\left(1-\beta\right)}{\left(1+2\beta\right)\left(1-\beta\phi\right)} \left(1+2\beta\phi+\left(K_{2}-3\beta^{2}\right)\phi^{2}\right)$$

added to the SPACE to take into account the effect of mass relocation on heat generation and thermal conductivity degradation.

Objectives of this paper

To implement the effective thermal conductivity model into the SPACE and verify the thermal conductivity degradation by using simple conceptual problem when fuel axial relocation in the ballooned fuel rods occurs.

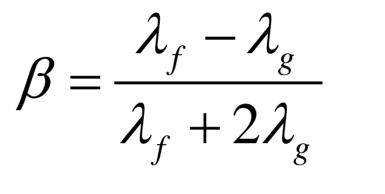
Strategy for implementation

- To calculate the fuel temperature in the SPACE, heat conduction equation has to be solved.
- When the fuel balloon or collapse is occurred, heat conduction equation can still be applied but thermal properties must be modified.

Before $\int \rho_f C_{pf} \frac{\partial T}{\partial t} = \int K_f \cdot \nabla T \, ds + \int P dV$

Where,

Thermal conductivity of fuel fragmentation (λ_f) , Thermal conductivity of gas (λ_{a}), and Packing fraction (ϕ)



 $K_2(\beta,\phi) \approx K_2^{(0)}(\beta) + K_2^{(1)}\phi$ $K_{2}^{(0)}(\beta) = 1.7383\beta^{3} + 2.8796\beta^{2} - 0.11604\beta$ $K_{2}^{(1)}(\beta) = 2.8341\beta^{3} - 0.13455\beta^{2} - 0.27858\beta$

Verification

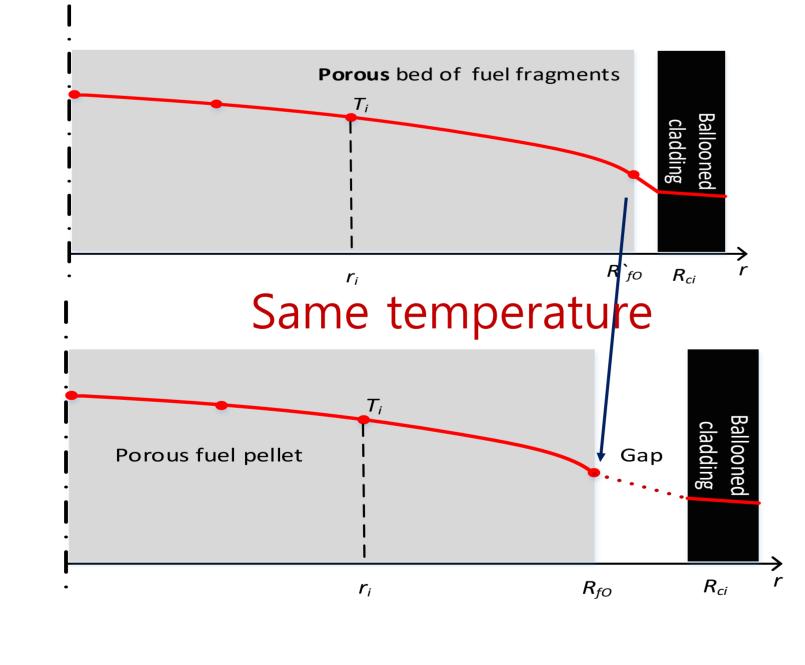
Conceptual problem is introduced with simple boundary condition. The SPACE results compares with analytic solutions.

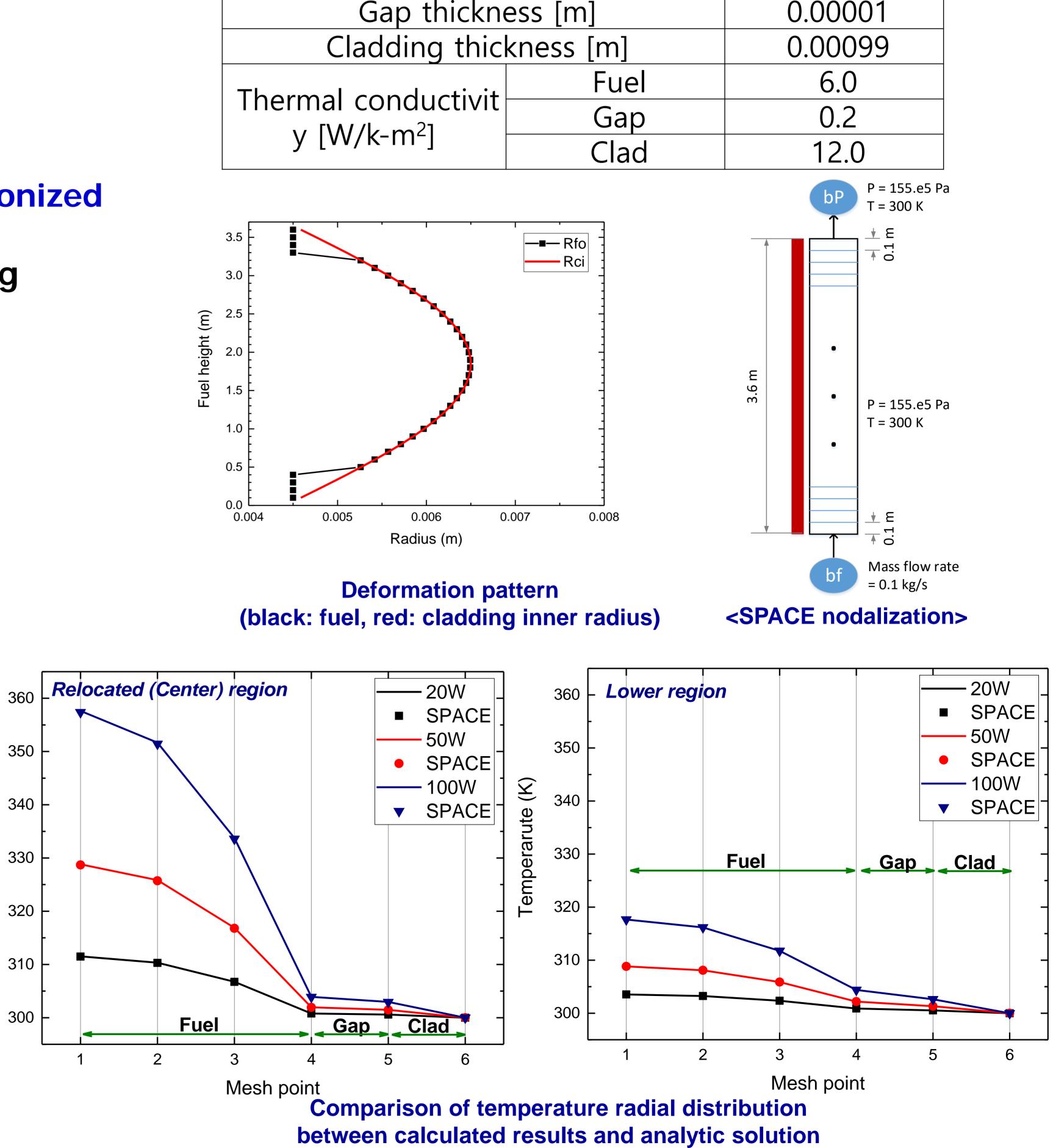
ICs & BCs

Properties	Value
Radius of pellet [m]	0.00450

After
$$\int_{V} \phi \rho_{f} C_{pf} \frac{\partial T}{\partial t} = \int_{s} K_{eff} \cdot \nabla T \, ds + \int_{V} \phi P dV$$

Temperature distributions need to be synchronized according to the fuel relocation because it is difficult to consider the changes of computing nodes due to the fuel ballooning in real time.





Conclusions

- nperarute (K) Effective thermal conductivity model is well implemented into the SPACE.
- The calculated results are well-matched with analytical solution.
- As a further work, we will simulate the **SET/IET to validate the newly implemented** models for FFRD phenomena.

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