

Compressor Off-design Performance Prediction with the Similitude Concept for Supercritical CO₂ Cooled Small Modular Reactor

Yongju Jeong^a, Seongmin Son^a, Seong Kuk Cho^a, Seungjoon Baik^b, Jeong Ik Lee^{a*}

^aDepartment of Nuclear and Quantum engineering, Korea Advanced Institute of Science and Technology, Korea

^bKorea Atomic Energy Research Institute, Daejeon, Korea

*Corresponding author: jeongiklee@kaist.ac.kr

1. Introduction

The demand for electricity has been steadily increasing over a century due to global industrial expansion. In the response to this trend, the capacities of power plants have been escalated. Subsequently, the researchers in nuclear power area have pursued the development of large size nuclear power plants. This movement resulted in centralization of power generation. However, nowadays, the market demands different kinds of characteristics from power plant. That is, a small scale power system with the feature of load following to accommodate the fluctuation in the electricity grid and diversify energy mix.

Therefore, nuclear industry came up with the idea of small modular reactor (SMR) and many researches are in progress. There are many issues to develop an SMR. One of the issues can be the selection of a power conversion cycle. Conventionally, the majority of nuclear power plants worldwide adopted a steam Rankin cycle, which has proven its efficiency and safety very well over the decades. However, because of its nature, the components of steam Rankine cycle tend to be large, and so is the system size.

One possible solution is to adopt the supercritical CO₂ (S-CO₂) Brayton cycle [1]. On the contrast to most Brayton cycles, the cycle operates with supercritical phase CO₂. S-CO₂ has liquid like density and gas like viscosity. Especially, if compression process occurs near the critical point, the compression work can be greatly reduced, due to the highly non-linear thermodynamics property changes including a reduced compressibility. Also, this cycle can deploy a compact turbomachinery from gas turbine technology. In other words, it may be possible to develop an SMR with high specific power and efficiency. Such example of an SMR using S-CO₂ power cycle is KAIST-MMR [2].

When the design of a system is complete, it is essential to observe the off-design performance for the safe operation of a nuclear power plant. For a system off-design analysis, component level off-design analysis is a prerequisite. One difficulty arises in an S-CO₂ compressor. Since the compressor works near the critical point, the validity of ideal gas base off-design performance model is in question. Hence, in this paper, the compressor off-design performance prediction methods are evaluated and the need to develop a new model for S-CO₂ compressors are proposed.

2. Methods and Results

Off-design operation of a compressor mostly means predicting the efficiency and pressure ratio with respect to off-design mass flow rate, rpm, inlet temperature and pressure. It might be very difficult or impractical to implement the iterations for four variables in a system code.

Instead, the concept of corrected mass flow rate and corrected rpm was originated from the similitude and widely utilized [3]. Specifically, compressor performances such as efficiency and pressure ratio are calculated according to various mass flow rates and rpms with the design point inlet temperature and pressure. To consider the varying inlet temperature and pressure ratio, similitude model converts those changes into mass flow rate and rpm. This is the concept of corrected mass flow rate and corrected rpm.

There exist several similitude models as shown in table 1 [4-6]. However, most of the models were proposed for air condition and not thoroughly evaluated for the S-CO₂ compressor application. To evaluate the accuracy of existing models, 1D-meanline code KAIST-TMD [7], developed by KAIST research team, was used. For comparison, performances of air and S-CO₂ compressors were calculated, and their results were summarized.

Table 1. Correction parameters for each model

	Mass parameter	rpm parameter	Head parameter
IG	$\frac{\dot{m}\sqrt{\gamma RT}}{\gamma P}$	$\frac{N}{\sqrt{\gamma RT}}$	$\frac{\Delta H}{\gamma RT}$
IGZ	$\frac{\dot{m}\sqrt{\gamma RZT}}{\gamma P}$	$\frac{N}{\sqrt{\gamma RZT}}$	$\frac{\Delta H}{\gamma RZT}$
BNI	$\frac{\dot{m}\sqrt{\gamma RZ_c T_c}}{\gamma P_c}$	$\frac{N}{\sqrt{\gamma RZ_c T_c}}$	$\frac{\Delta H}{\gamma RZ_c T_c}$
Glassman	$\frac{\dot{m}V_{cr}}{P\gamma\left(\frac{2}{\gamma+1}\right)^{\gamma/(\gamma-1)}}$	$\frac{N}{V_{cr}}$	$\frac{\Delta H}{V_{cr}^2}$
CEA	$\frac{\dot{m}\sqrt{n_s RZT}}{n_s P}$	$\frac{N}{\sqrt{n_s RZT}}$	$\frac{\Delta H}{n_s RZT}$

2.1 S-CO₂ compressor case

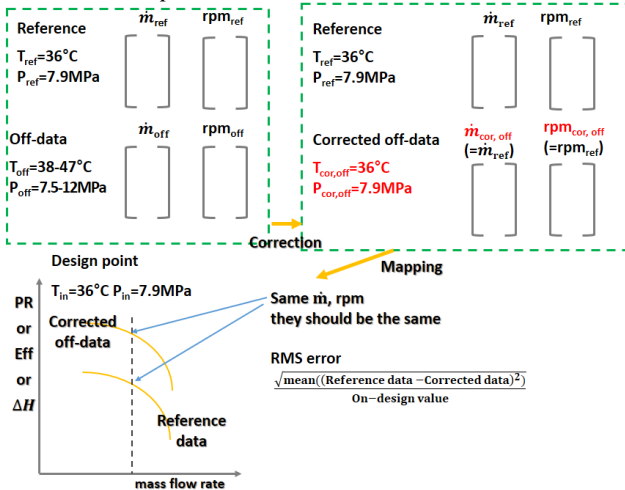


Figure 1. Correction and mapping procedure diagram

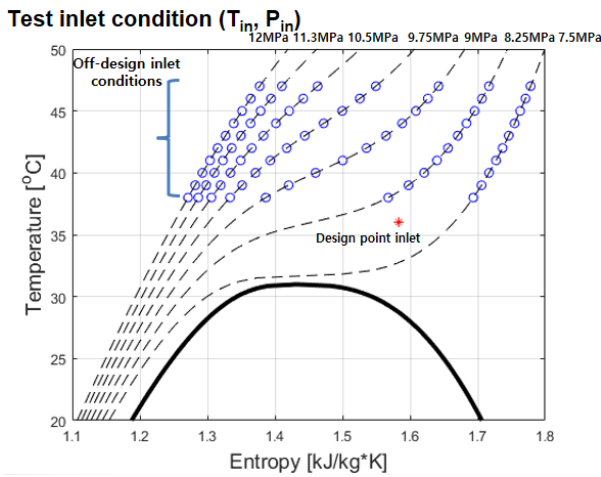


Figure 2. Test off-design inlet condition

Table 2. Design point inlet condition

Design point			
T (°C)	36	ρ (kg/m ³)	329.4
P (MPa)	7.92	γ	9.0
m (kg/s)	129.14	Z	0.41
rpm (rev/min)	15000	n_s	1.54

Table 3. Range of studied inlet condition

	min	max	resolution
T (°C)	38	47	10
P (MPa)	7.5	12	7
rpm (rev/min)	4500	16500	9

To evaluate the accuracy of correction models, an example compressor was designed first and its performance was calculated with respect to various mass flow rate and rpm, but with design inlet temperature and pressure. Next, arbitrary off-design inlet conditions (temperature, pressure) were prescribed as shown in Fig 2 and Table 3. Additionally, their performance were

calculated so that the off-design performance can be corrected into the design inlet condition (temperature, pressure). The detailed correction procedure is shown in Fig 1. The implication of this method is that it is possible to predict the compressor performance at design inlet temperature and pressure with off-design inlet temperature and pressure data. However, since the similitude does not always hold perfectly, discrepancies occur. For example, in Figs. 3 and 4, the red lines indicate the performance of design point inlet condition data, and the black lines indicate the off-design inlet condition data which originally has 45°C and 9MPa, but converted into 36°C and 7.9MPa to predict the design point performance. Because of the imperfect similitude, the error occurs, otherwise the red and black lines should be overlapped precisely.

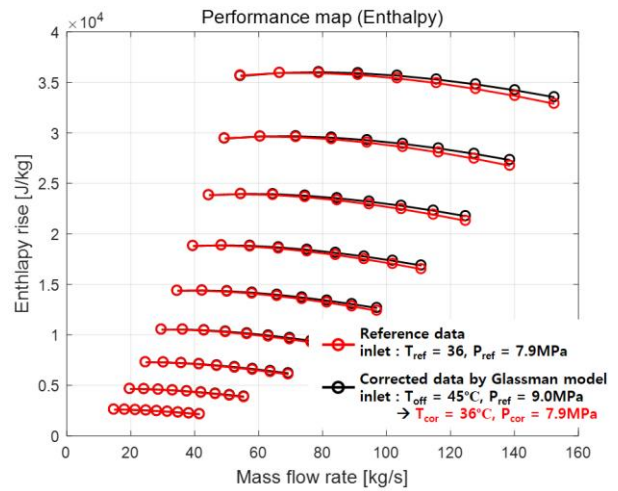


Figure 3. Performance comparison example (Enthalpy rise)

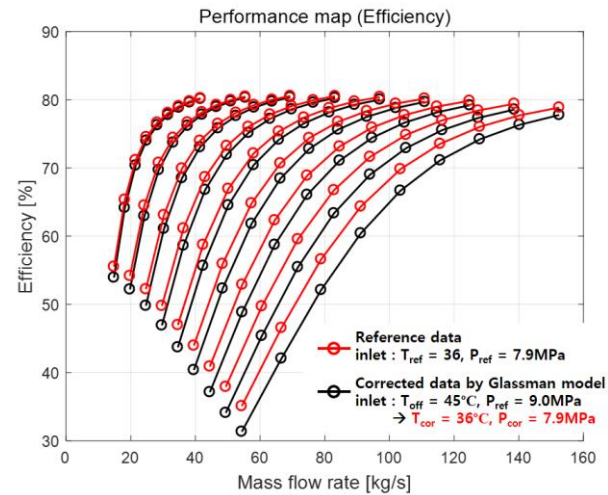


Figure 4. Performance comparison example (Efficiency)

As mentioned above, there are several models for correction. Thus, for the inlet conditions in Table 3, the discrepancies were quantified as a normalized RMS. The results were summarized in Table 4. Overall, the prediction using enthalpy rise showed more accurate prediction than pressure ratio. Especially, IGZ and CEA

models have superior accuracy. Efficiency prediction has approximately 10% errors, which is higher error than enthalpy rise.

Table 4. Performance discrepancy error (normalized RMS error) summary (unit : [%])

	<i>PR error</i>	<i>Eff error</i>	ΔH error	<i>PR error (PR by ΔH)</i>
<i>IG</i>	19.7	13.6	7.16	3.46
<i>IGZ</i>	20.2	8.8	2.04	1.24
<i>BNI</i>	8.24	12.5	10.1	3.56
<i>Glassman</i>	9.9	15.1	9.04	3.83
<i>CEA</i>	48.0	9.63	2.83	1.27

2.2 Air compressor case

Table 5. Design point inlet condition

<i>Design point</i>			
<i>T (°C)</i>	20.3	ρ (kg/m ³)	1.266
<i>P (MPa)</i>	7	γ	1.4
<i>m (kg/s)</i>	129.14	<i>Z</i>	1
<i>rpm (rev/min)</i>	15000	<i>n_s</i>	1.4

Table 6. Range of studied inlet condition

	<i>min</i>	<i>max</i>	<i>resolution</i>
<i>T (°C)</i>	15	40	6
<i>P (MPa)</i>	75	250	8
<i>rpm (rev/min)</i>	3300	12100	9

Table 7. Performance discrepancy error (normalized RMS error) summary (unit : [%])

	<i>PR error</i>	<i>Eff error</i>	ΔH error	<i>PR error (PR by ΔH)</i>
<i>IG</i>	0.493	0.778	0.753	0.459
<i>IGZ</i>	0.522	0.773	0.805	0.493
<i>BNI</i>	0.462	0.773	0.747	0.456
<i>Glassman</i>	0.471	0.782	0.750	0.457
<i>CEA</i>	0.475	0.774	0.750	0.457

To compare the correction model accuracy for S-CO₂ and air environment, the same procedure for an air compressor was performed as in section 2.1. Design and off-design conditions were tabulated in Tables 5 and 6. The results are summarized in Table 7. The air compressor results showed less than 1% error for all indicators. Noticeably, regardless of the models, the results displayed almost the same accuracy.

3. Summary and Conclusions

To ensure the safe operation of an SMR with the S-CO₂ power cycle, the compressor off-design performance prediction method derived from similitude analysis was investigated. Among existing models, five

models were selected and their prediction accuracies were tested with 1D meanline code, KAIST-TMD. For comparison, both air and S-CO₂ cases were calculated. In case of an air compressor, all the correction methods show an equal level of accuracies with high fidelity to the direct performance prediction. However, the errors for an S-CO₂ case were uneven up to the similitude models. Moreover, the overall values are noticeably larger than the air case. In conclusion, applying the existing similitude correction models, developed under air conditions, to the off-design performance prediction of an S-CO₂ compressor may result in significant errors. Consequently, the cause of errors should be studied and a new model needs to be developed for an S-CO₂ compressor.

4. Acknowledgement

This research was supported by Civil-Military Technology Cooperation Program (iCMTTC) funded by the Agency for Defense Development – South Korea (17-CM-EN-04).

REFERENCES

- [1] Ahn, Y., Bae, S. J., Kim, M., Cho, S. K., Baik, S., Lee, J. I., & Cha, J. E. (2015). Review of supercritical CO₂ power cycle technology and current status of research and development. *Nuclear Engineering and Technology*, 47(6), 647-661.
- [2] Kim, S. G., Yu, H., Moon, J., Baik, S., Kim, Y., Jeong, Y. H., & Lee, J. I. (2017). A concept design of supercritical CO₂ cooled SMR operating at isolated microgrid region. *International Journal of Energy Research*, 41(4), 512-525.
- [3] Saravanamuttoo, H. I., Rogers, G. F. C., & Cohen, H. (2001). *Gas turbine theory*. Pearson Education.
- [4] Roberts, S. K., & Sjolander, S. A. (2005). Effect of the specific heat ratio on the aerodynamic performance of turbomachinery. *Journal of engineering for gas turbines and power*, 127(4), 773-780.
- [5] Glassman, A. J. (1972). *Turbine design and application*. nasa sp-290. NASA Special Publication, 290.
- [6] Pham, H. S., Alpy, N., Ferrasse, J. H., Boutin, O., Tothill, M., Quenaut, J., ... & Saez, M. (2016). An approach for establishing the performance maps of the sc-CO₂ compressor: Development and qualification by means of CFD simulations. *International Journal of Heat and Fluid Flow*, 61, 379-394.
- [7] Lee, J., 2016. Study of improved design methodology of S-CO₂ power cycle compressor for the next generation nuclear system application, Thesis, KAIST.