

Buckling Analyses and Tests for Thin Cylindrical Structures
Subjecting the Shear Loads

150

. J. Okada

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L/R=3.1, 1.6, 1.0

가

ANSYS

L/R=3

, L/R=1.0

, L/R=1.6

ABSTRACT

In this paper, the buckling analyses and the tests for thin cylindrical structures subjecting shear load are carried out to develop the buckling design and analysis technology for liquid metal reactor. Using the buckling strength formulae proposed by J. Okada, the reduced test specimen were designed and fabricated, which have the slenderness ratio $L/R=3.1$, 1.6 and 1.0 . For the test specimen, the buckling analyses using ANSYS and the tests using the hydraulic actuator are performed. From the comparison of the buckling loads between analysis and test, the analysis results are in good agreement with those of tests. The larger slenderness ratio over $L/R=3$ results in dominant bending buckling mode, smaller slenderness ratio under $L/R=1$ results in dominant shear buckling mode, and near $L/R=1.6$ region shows the mixed buckling mode which has the bending and shear buckling mode simultaneously.

1.

500°C
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5

[1].

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[2,3].

(L/R),

(buckling Load Limit Load)

[3,4,5].

Okada

가 [2]

가 3 가

ANSYS

가

2.

2.1

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Fig. 1 가

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L/R=3.1(Model-

A),

L/R=1.6(Model-B),

L/R=1.0(Model-C) 3 가

. Table 1

t=0.5mm

D=103mm

304

Fig. 2

1

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2.5 ,

±12.5cm

Shenck

Labtronic 8800

가 가

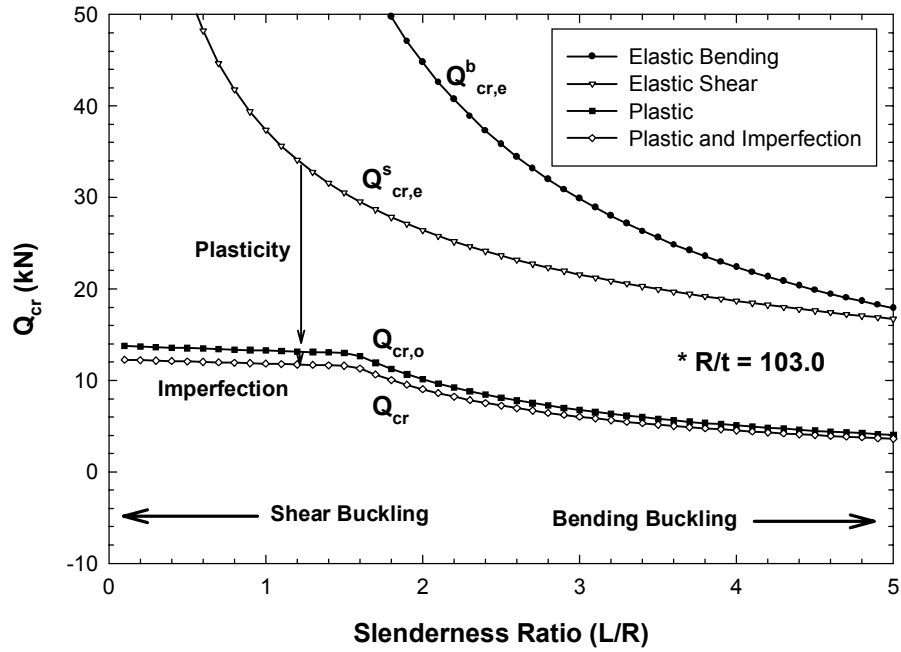


Fig. 1 Results of Buckling Strength Evaluation by Formulae ($t=0.5\text{mm}$, $D=103\text{mm}$, $L=103\text{mm}$)

Table 1. Dimensions of Buckling Test Specimen

	Model-A (Bending)	Model-B (Bending + Shear)	Model-C (Shear)
L(mm)	160.0	80.0	50.6
D(mm)	103.0	103.0	103.0
t(mm)	0.5	0.5	0.5
(L/R)	3.1	1.6	1.0

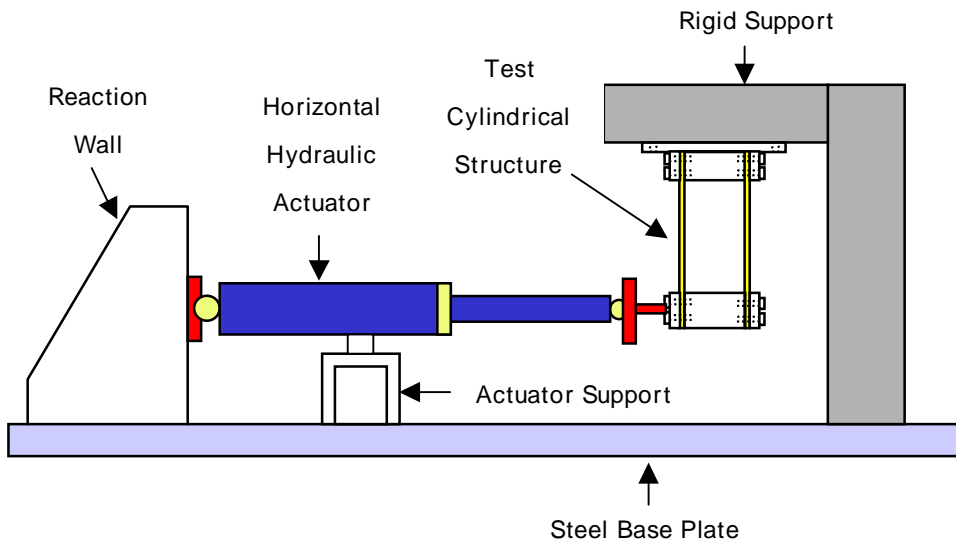


Fig. 2 Schematic Drawing of 1-Directional Buckling Test facility

2.2

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ANSYS

[6]

4

SHELL43

E=194GPa,

$\nu=0.29$, $\rho=7800\text{kg/m}^3$

Fig. 3

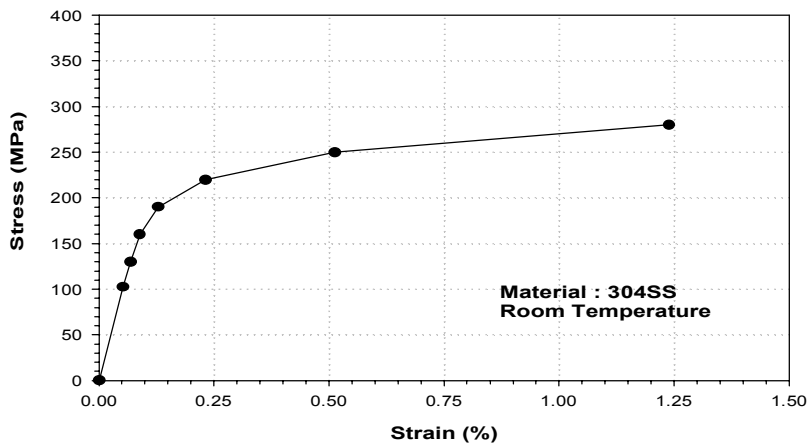


Fig. 3 Stress-Strain Curve Used in Nonlinear Elastic-Plastic Buckling Analysis

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L/R=3.1

Model-A

Fig. 4

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Fig. 5

Model-A

$Q_{cr} = 7.6\text{kN}$,

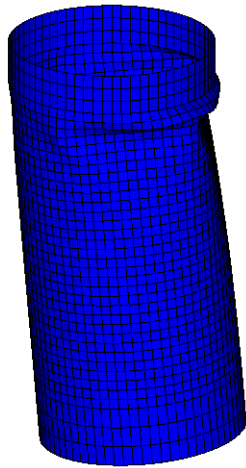
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$Q_{cr} = 6\text{kN}$,

가

$Q_{cr} = 7\text{kN}$

가



(Analysis)



(Test)

Fig. 4 Buckling Shape of Mode-A Specimen

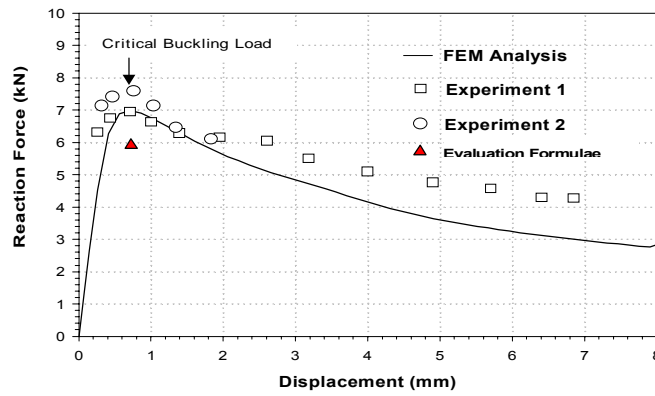


Fig. 5 Load-Displacement Curves for Model-A Specimen

Fig. 6
B

L/R=1.6
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Model-

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=10.8kN,

$Q_{cr} = 11\text{kN}$

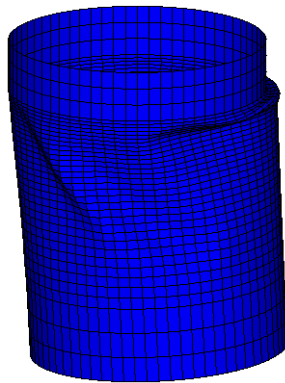
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Fig. 7 Model-B

$Q_{cr} = 12\text{kN}$,

가

Q_{cr}



(Analysis)



(Test)

Fig. 6 Buckling Shape of Model-B Specimen

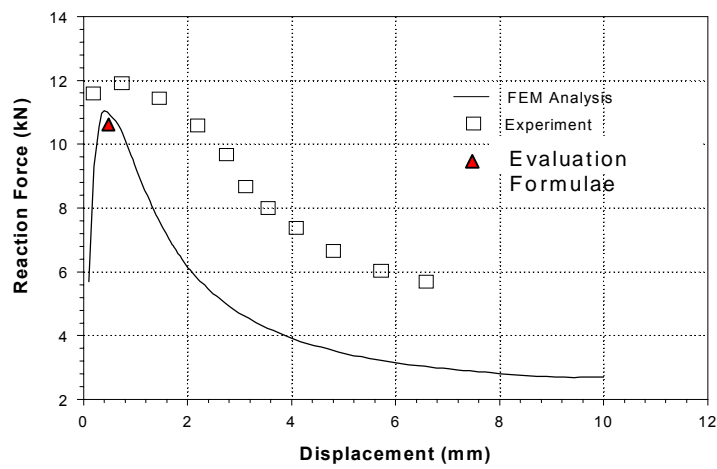


Fig. 7 Load-Displacement Curves for Model-B Specimen

Fig. 8

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Fig. 9

$Q_{cr} = 17.5\text{kN}$,

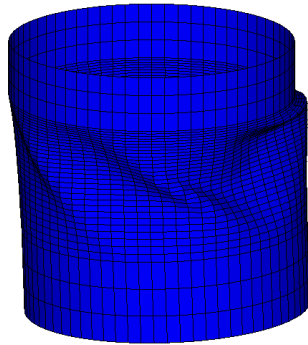
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$Q_{cr} = 11\text{kN}$,

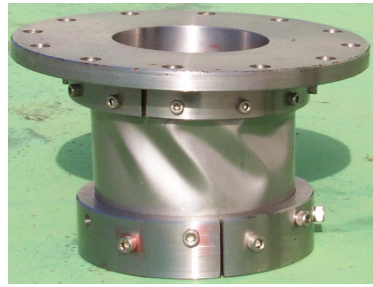
Q_{cr}

=12.2kN

가 가



(Analysis)



(Test)

Fig. 8 Buckling Shape of Model-C Specimen

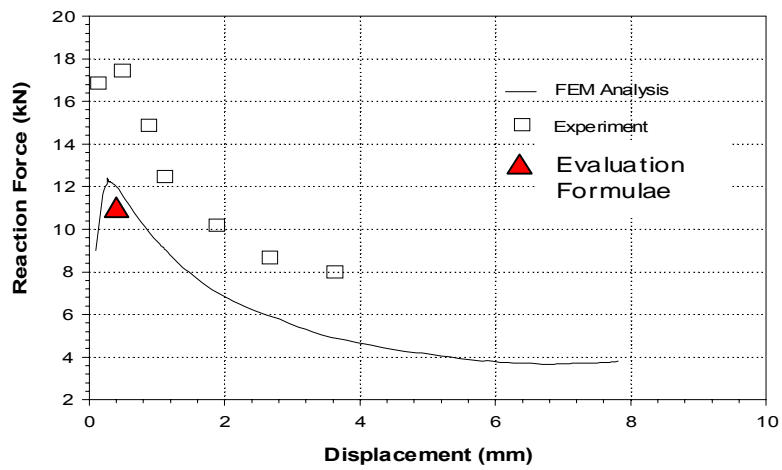


Fig. 9 Load-Displacement Curves for Model-C Specimen

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Fig. 10

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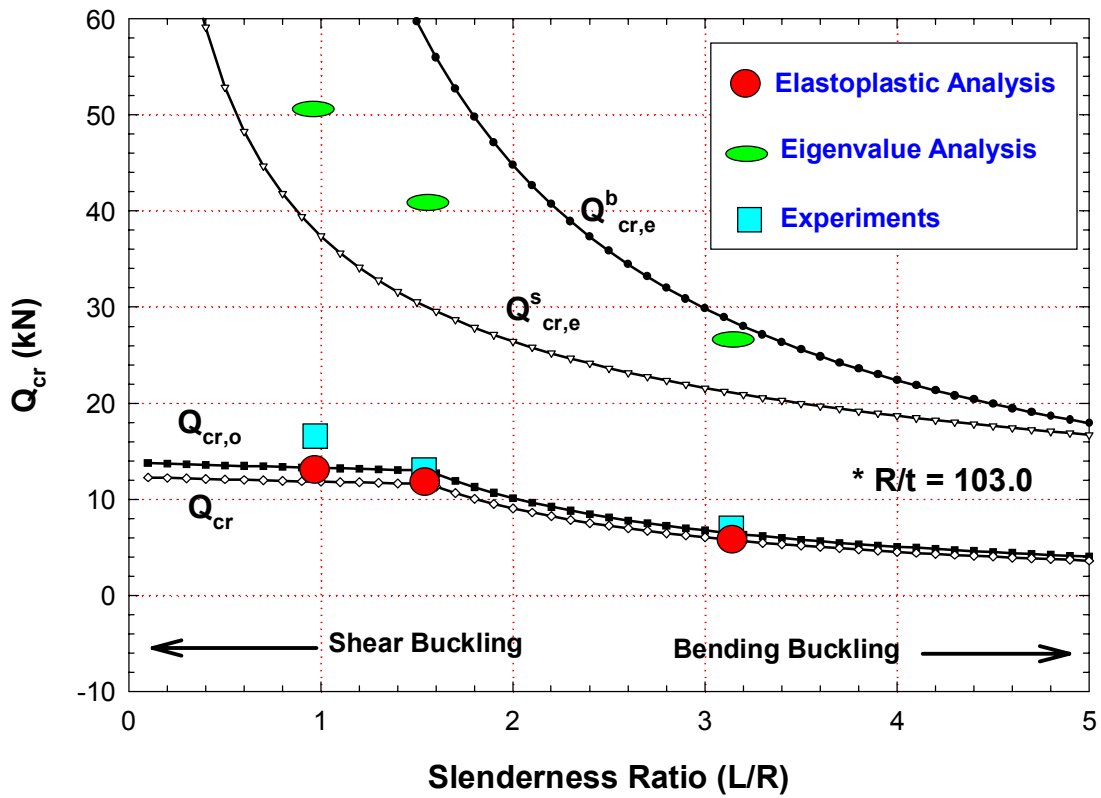


Fig. 10 Comparison of Buckling Strength Values Corresponding to Slenderness Ratio by Formulae, Analyses and Tests

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