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SMART-P 증기발생기 유동분배판에 대한 전산유체해석

CFD analysis on the flow distributing plate of SMART-P SG

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Abstract

Flow distributing plates are installed in the Steam Generator(SG) upper annular of SMART-P. The current flow distribution plates changes the direction of MCP discharge flow from vertical to horizontal. The flow balance on the SG cassettes are made by SG orifice only. However, it couldn't get uniform flow distribution on each SG cassette. To overcome this drawbacks, design concept of improving flow distribution on the SG cassettes is introduced to the ring-shaped perforated plate instead of the current distributing plates. In this paper the related effects between SG orifice and flow distribution using a developed 2D CFD code. RNG k-e model, 2nd Upwind scheme, staggered grid, and SIMPLE algorithm was adopted in the 2D CFD code to analyze this model. The maximum flow distribution error on the SG cassettes was reduced to about 0.1%. The pressure loss from the SG header to SG orifice was also reduced about 30%. As a results, much improvement was made in flow balancing on the SG cassettes and in the pressure loss reduction by introducing the ring-shaped perforated plate.

1.

SMART -P

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, 4 (train) .

. SMART -P () 가 • 가 가 가 가 가 가 가 가 가 가 . 가 가 . 가 가 가 MCP가 가 가 SMART -P .

가 가 SMART-P MCP

(2(a)) 1 (2(b)) , 가

> (analytical) 가 , ^[1~4]. 가

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2. 2.1

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$$div(\mathbf{u}) = 0 \tag{1}$$

$$x \qquad : \ \rho \frac{\partial u}{\partial t} + \rho \ div (u \mathbf{u}) = -\frac{\partial p}{\partial x} + \mu \ div \ grad (u)$$
(2)

$$y : \rho \frac{\partial v}{\partial t} + \rho \, div \, (v \, \mathbf{u}) = -\frac{\partial p}{\partial y} + \mu \, div \, grad \, (v)$$
(Navier - Stokes equation)

(Reynolds -averaged)

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$$\mathbf{U} = \frac{1}{t_1 - t_0} \int_{t_0}^{t_1} \mathbf{u}(t) dt$$
(3)

$$\mathbf{u} = \mathbf{U} + \mathbf{u}' \tag{4}$$

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(4) (1), (2) **u**' 0 (1), (2) (time averaged) 2

$$div(\mathbf{U}) = 0 \tag{5}$$

$$x : \rho \frac{\partial U}{\partial t} + \rho \operatorname{div} (U\mathbf{U}) = -\frac{\partial P}{\partial x} + \mu \operatorname{div} \operatorname{grad} (U) - \rho \left(\frac{\partial \overline{u'^2}}{\partial x} + \frac{\partial \overline{u'v'}}{\partial y} \right)$$
(6)

$$y : \rho \frac{\partial V}{\partial t} + \rho \operatorname{div} (V\mathbf{U}) = -\frac{\partial P}{\partial y} + \mu \operatorname{div} \operatorname{grad} (V) - \rho \left(\frac{\partial \overline{v'^2}}{\partial y} + \frac{\partial \overline{u'v'}}{\partial x} \right)$$
(2) (6) $-\rho \overline{u_i u_j}$.

2.2 RNG
$$k-\varepsilon$$
 ^[5-8] 7

$$\rho \,\overline{u_i u_j} = \frac{2}{3} \,\rho k \delta_{ij} - \mu \, \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \tag{7}$$

2

RNG
$$k-\varepsilon$$
 (k) $k-\epsilon$, (ϵ) $k-\epsilon$ $(-\rho R)$ 7^{+} . 2,,RNG $k-\varepsilon$ k, ϵ .

i k, ϵ

$$\rho U_{i} \frac{\partial k}{\partial x_{i}} = \alpha_{k} \mu_{eff} \frac{\partial^{2} k}{\partial x_{i}^{2}} + 2\mu_{t} E_{ij} \cdot E_{ij} - \rho \varepsilon$$
(8)

$$\rho U_{i} \frac{\partial \varepsilon}{\partial x_{i}} = \alpha_{\varepsilon} \mu_{eff} \frac{\partial^{2} \varepsilon}{\partial x_{i}^{2}} + C_{1} \frac{\varepsilon}{k} 2 \mu_{t} E_{ij} \cdot E_{ij} - C_{2} \rho \frac{\varepsilon^{2}}{k} - R$$
(9)

$$\mu_{eff} = \mu + \mu_t \tag{10}$$

$$\mu_t = \rho c_\mu \frac{k^2}{\varepsilon} \tag{11}$$

RNG
$$k - \epsilon$$
 (μ_t) $k - \epsilon$ C_{μ}

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, Reynolds

$$\mu_{t} = \mu \left[1 + \sqrt{\frac{C_{\mu}}{\mu}} \frac{k}{\sqrt{\varepsilon}} \right]^{2}$$
(12)

RNG $k - \epsilon$ R

$$R = C_{\mu}\rho \frac{\eta^{3}(1 - \eta/\eta_{0})}{1 + \beta\eta^{3}} \frac{\varepsilon^{2}}{k}$$
(13)

$$\eta = \sqrt{\left(2S_{ij} \cdot S_{ij}\right)} \frac{k}{\varepsilon} \tag{14}$$

$$\beta = 0.012$$

$$\eta_0 = \sqrt{\frac{C_2 - 1}{C_\mu (C_1 - 1)}}$$

2

$$S_{ij} = \frac{1}{2} \left(\frac{\partial U}{\partial y} + \frac{\partial V}{\partial x} \right)$$
$$S_{ij} \cdot S_{ij} = \left(\frac{\partial U}{\partial x} \right)^2 + \left(\frac{\partial V}{\partial y} \right)^2 + \frac{1}{2} \left[\left(\frac{\partial U}{\partial y} \right) + \left(\frac{\partial V}{\partial x} \right) \right]^2$$

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RNG
$$k-\varepsilon$$

C_{μ}	C_1	C_2	α_k	${\mathcal A}_{arepsilon}$
0.0845	1.42	1.68	1.39	1.39

(rate of strain) , $\eta (= Sk/\epsilon)$ RNG $k-\varepsilon$ R, $S^2 (= 2S_{ij} \cdot S_{ij})$. $k-\varepsilon$. 가 . RNG $k-\varepsilon$ (13) η η_0 C_2 가 η アト η_0 R η フト η_0 가 . . C_2 가 C_2 R(ε) 가 . RNG $k-\varepsilon$ $k-\varepsilon$. 4. $\mathrm{RNG} \ k-\epsilon$ 2 (staggered) , MCP 2 , $k,\;\epsilon$ 3). SMART -P [18] (• 1 1/4 . 3 MCP Euler Number($E = \rho v 2 / \Delta p$) . SG 1 4 2 MCP 가 . 2 6

가

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나 4 가 2 가

3

(*Q_i*) 가

2

 $\theta_{diff}(i) = \frac{Q_i - Q_{ideal}}{Q_{ideal}} \times 100 [\%]$ $\theta_{\max} = |\theta_{diff \max} - \theta_{diff \min}| [\%]$ $Q_{ideal} = \frac{Q_{total}}{N_{SG}}$ (15)

4(a)

(b)



표 1. 격자민감도, 입구유동, 출구영향 조사 해석Case

Case	P [KPa]	SG+Orifice P [KPa]		
A1	1.0	10	А	271×169
A2	1.0	10	А	406×169
A3	1.0	10	А	406×224
A4	1.0	10	А	406×317
B1	0.1	10	А	
B2	0.1	10	А	
C1	0.1	10	А	
C2	0.1	10	А	
C3	0.1	10	А	
C4	0.1	10	A	

표 2. 격자민감도, 입구유동, 출구영향 해석결과

	Case	Q_i/Q_{ideal} [%]				
		SG1	SG2	SG3		
	A1	99.989	100.007	100.004	A	271×169
	A2	99.988	100.006	100.006	A	406×169
	A3	99.989	100.007	100.004	A	406×224
	A4	99.989	100.006	100.005	A	406×317
	B1	100.10	99.83	100.07	A	
	B2	100.11	99.82	100.06	A	
	C1	100.09	99.84	100.07	A*	
	C2	100.08	99.87	100.05	A*	
	C3	100.10	99.83	100.07	A [*]	
	C4	100.10	99.84	100.06	A*	



(heta max)가

표 3. 설계변수 해석 Case

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	Reynolds ($ ho v H/\mu$)	Case	P [KPa]	SG+Orifice P [KPa]	
	2.04E6	D1	0	15	A**
	1.02E6	D2	0	15	A**
	2.04E6	D3	0	20	A**
	1.02E6	D4	0	20	A**
	2.04E6	E1	5 -0.01	10~14.99	А
	2.04E6	E2	5 -0.01	10~14.99	В
()	2.04E6	E3	5 -0.01	10	A
	2.04E6	E4	5 -0.01	10	В
	1.02E6	E5	5 -0.01	10	A

표 4. 증기발생기 유량 및 최대유량편차(기존설계)

	Case	Q_i / Q_{ideal} [%]			A	
		SG1	SG2	SG3	<i>diff</i> max	
	D1	99.82	98.68	101.50	2.82	А
	D2	99.84	98.68	101.48	2.80	А
	D3	99.75	98.35	101.90	3.55	А
	D4	99.76	98.35	101.89	3.53	А

Case E () (perforated plates) . 15KPa 가 Case E1 E2 0.01KPa 5KPa 가 10KPa 14.99 KPa . 8 Case E1, E2 (θ_{max}) . 가

.

가 0.5% Case E1 50Pa 10 Pa~1 KPa . 가 가 12). (가 1KPa . Case E2 가 0.5% 200Pa 가 Case E1 . . Case E2 가 1KPa Case E1 .

 Case E3
 E4
 10KPa

 5KPa
 0.01KPa
 .
 9
 Case

 E3, E4
 (θ_{max})
 .
 Case E3, E4
 Case E3, E4

 Case E1, E2
 ?
 ?
 ?

Case E3 75Pa 가 0.5% 10 Pa~1 KPa Case E1 . 1KPa Case E1, E2 가 . Case E4 300Pa 가 0.5% 가 Case E2 Case E1, E3 . Case E4 1KPa 가 Case E1~E3 •

가 0.5% 50Pa Case E1 14.95KPa, Case E2 200Pa 14.80KPa, 75Pa Case E3 10KPa, Case E4 300Pa 10KPa Case E1 Case E3 4.925 KPa, Case E2 Case E47 4.7 KPa KPa Ра ~

가 ., . 10(a), (b) Case E1, E2 Case E3, E4 . 가 . Reynolds 가 11 . 가 50Pa 6. RNG $k-\epsilon$, 2 , (staggered) , SIMPLE 2 . SMART -P MCP 가. (1/12 (case D1~D4) 2) (heta $_{
m max}$) 15 KPa 3.5 %, 20 2.8% KPa . , 3 가 가 () (.) 300Pa 가 0.5% 가 ,1KPa . () 1KPa . 10KPa () 가 , () 3.5~2.8% 0.1% 15KPa 11KPa 4KPa . • , MCP 가 가 .





MCP/SG header section







Fig. 3. Inlet U, V, k, ϵ Profile



 $(b) \quad Type \ B \label{eq:type}$ Fig. 4. MCP 2D analysis model







Fig. 6. Velocity of header section between SG1 and SG2(Grid Sensitivity)



Fig. 7. 2D analysis model of the existing MCP header



Fig. 10. Variation with the orifice $\Delta {\bf P}$ of maximum flow rate diff.(θ $_{\rm max}$)



Fig. 11. Variation with inlet Reynolds no. of maximum flow rate diff. ($\theta_{\rm max}$)



Fig. 12. Total pressure distribution contours of case E3

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