DVI

Development of Analytical Model for the Direct ECC Bypass in DVI System Downcomer



Abstract

An analytical model for the direct ECC bypass is developed which can predict the bypass fraction. It is derived from continuity equation of liquid phase and two dimensional force balances between liquid and gas. The bypass rate can be calculated with the liquid and gas injection conditions and constitutive relations of wall and interfacial friction factors. Some experimental results of direct ECC bypass are compared with the proposed model to validate it. Also, the sensitivity studies about interfacial friction factor and liquid spreading width which are the important factors for the direct ECC bypass are conducted. The proposed model shows qualitatively and quantitatively successful results if the data of liquid spreading width and onset of sweep-out are reflected in the analytical model.

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Wilcock) (DVI) , , , , , , , , , , , , , , , , , , , , , , , , , , , , , , , , , ,
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21 , sweep-out , UPTF , 1 , DIVA MIDAS APR-1400 UPTF (4.5). . . .<
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[4,5]. [6] 1/10 APR-1400 [7], [8]. [9,10] (analytical model) Wallis ⁷ , [11]. , 2
[4,5]. , [6] 1/10 APR-1400 [7], [8]. [9,10] (analytical model) . 2 , 2 Wallis7 , 2 [11].
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[7], [8]. [9,10]
[8]. [9,10]
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, Wallis7⊦ [11]. , 2
Wallis7† [11].
, 2
(void
fraction) . ,
DIVA APR-1400 ,
(interfactal interior)

1.









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2. DVI-4



(a) Impinging jet3. DVI-2



2 DVI-4 DVI . 가 가 . DVI-4 DVI-2 가 . DVI-4 impinging jet 3-(a) DVI-2 . -2 -3 , . impinging jet impinging jet . , • 가 3-(b) (cross flow) 2 , , . -1 가 DVI-2 . 가 . 3. . 가 . • 가 . separated flow (film flow annular flow) . , annular flow film flow . 4 .







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1) Film flow

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$$-\Delta P \cdot D_1 \cdot W + \tau_{W\!f\!y} \cdot H \cdot W + \tau_{iy} \cdot H \cdot W - \rho_f \cdot g \cdot D_1 \cdot W \cdot H = 0 \quad \text{or}$$

$$-\frac{\Delta P}{H} + \frac{\tau_{Wfy}}{D_1} + \frac{\tau_{iy}}{D_1} - \rho_f g = 0$$
(1)

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$$-\Delta P \cdot D_2 \cdot W + \tau_{Wgy} \cdot H \cdot W - \tau_{iy} \cdot H \cdot W - \rho_g \cdot g \cdot D_2 \cdot W \cdot H = 0 \quad \text{or}$$

$$-\frac{\Delta P}{H} + \frac{\tau_{Wgy}}{D_2} - \frac{\tau_{iy}}{D_2} - \rho_g g = 0$$
(2)

(1) (2)

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$$\left(\frac{\tau_{Wfy}}{D_1} - \frac{\tau_{Wgy}}{D_2}\right) + \left(\frac{\tau_{iy}}{D_1} + \frac{\tau_{iy}}{D_2}\right) = (\rho_f - \rho_g)g \quad \text{or} \quad \left[\frac{\tau_{Wfy}}{(1-\alpha)} - \frac{\tau_{Wgy}}{\alpha}\right] + \left[\frac{1}{(1-\alpha)} + \frac{1}{\alpha}\right] \cdot \tau_{iy} = D \cdot (\rho_f - \rho_g)g \quad (3)$$

$$\therefore \quad \frac{D_1}{D} = (1 - \alpha) \text{ and } \quad \frac{D_2}{D} = \alpha$$
(4)

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$$-\Delta P \cdot D_1 \cdot H - \tau_{Wfx} \cdot H \cdot W + \tau_{ix} \cdot H \cdot W = 0 \quad \text{or} \qquad -\frac{\Delta P}{W} - \frac{\tau_{Wfx}}{D_1} + \frac{\tau_{ix}}{D_1} = 0$$
(5)

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$$-\Delta P \cdot D_2 \cdot H - \tau_{Wgx} \cdot H \cdot W - \tau_{iy} \cdot H \cdot W = 0 \quad \text{or} \quad -\frac{\Delta P}{W} - \frac{\tau_{Wgx}}{D_2} - \frac{\tau_{ix}}{D_2} = 0 \tag{6}$$

(5)

(6)

$$\left(\frac{\tau_{Wfx}}{D_1} - \frac{\tau_{Wgx}}{D_2}\right) - \left(\frac{\tau_{ix}}{D_1} + \frac{\tau_{ix}}{D_2}\right) = 0 \quad \text{or} \quad \left(\frac{\tau_{Wfx}}{1 - \alpha} - \frac{\tau_{Wgx}}{\alpha}\right) - \left(\frac{\tau_{ix}}{1 - \alpha} + \frac{\tau_{ix}}{\alpha}\right) = 0 \tag{7}$$

$$\tau_{Wfy} = \frac{1}{2} \rho_f f_{Wfy} \upsilon_f | \vec{u}_f | = \frac{1}{2} \rho_f f_{Wfy} \upsilon_f \sqrt{u_f^2 + \upsilon_f^2} , \quad \tau_{Wgy} = \frac{1}{2} \rho_g f_{Wgy} \upsilon_g \sqrt{u_g^2 + \upsilon_g^2} ,$$

$$\tau_{Wfx} = \frac{1}{2} \rho_x f_{Wfx} u_f \sqrt{u_f^2 + \upsilon_f^2} , \quad \tau_{Wgx} = \frac{1}{2} \rho_g f_{Wgx} u_g \sqrt{u_g^2 + \upsilon_g^2}$$

$$\upsilon_g = 0 \qquad 7! \qquad (8)$$

$$\tau_{Wgx} = \frac{1}{2} \rho_g f_{Wgx} u_g^2, \qquad \tau_{Wgy} = 0$$
(9)

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x y (10) [13].

$$\tau_{iy} = \frac{1}{2} \rho_g f_{iy} | \stackrel{\mathbf{r}}{u_g} - \stackrel{\mathbf{r}}{u_f} | (\upsilon_g - (-\upsilon_f)) , \quad \tau_{ix} = \frac{1}{2} \rho_g f_{ix} | \stackrel{\mathbf{r}}{u_g} - \stackrel{\mathbf{r}}{u_f} | (u_g - u_f)$$
(10)

[13].

(10)

$$, \quad | \stackrel{\mathbf{f}}{u_g} - \stackrel{\mathbf{f}}{u_f} | = \sqrt{\left(u_g - u_f\right)^2 + \left(\upsilon_g - \upsilon_f\right)^2} , \quad (10) \qquad .$$
$$\tau_{iy} = \frac{1}{2} \rho_g f_{iy} \sqrt{\left(u_g - u_f\right)^2 + \upsilon_f^2} (\upsilon_f) , \\ \tau_{ix} = \frac{1}{2} \rho_g f_{ix} \sqrt{\left(u_g - u_f\right)^2 + \upsilon_f^2} (\upsilon_g) , \quad (11)$$

(8), (9) (11) (3) (7)

$$- y$$

$$\frac{\frac{1}{2}\rho_{f}f_{Wfy}\upsilon_{f}\sqrt{u_{f}^{2}+\upsilon_{f}^{2}}}{(1-\alpha)} + \left[\frac{1}{(1-\alpha)} + \frac{1}{\alpha}\right] \cdot \frac{1}{2}\rho_{g}f_{iy}\sqrt{(u_{g}-u_{f})^{2}+\upsilon_{f}^{2}}(\upsilon_{f}) = D \cdot (\rho_{f}-\rho_{g})g \qquad (12)$$

$$- x -$$

$$\frac{\frac{1}{2}\rho_f f_{Wfx} u_f \sqrt{u_f^2 + v_f^2}}{(1 - \alpha)} - \frac{\frac{1}{2}\rho_g f_{Wgx} u_g^2}{\alpha} - \left[\frac{1}{(1 - \alpha)} + \frac{1}{\alpha}\right] \cdot \frac{1}{2}\rho_g f_{ix} \sqrt{(u_g - u_f)^2 + v_f^2} (u_g - u_f) = 0$$
(13)

$$j_{kx}^* = \alpha_k u_k \cdot \sqrt{\frac{\rho_k}{D \cdot (\rho_f - \rho_g) \cdot g}} \qquad \qquad j_{ky}^* = \alpha_k v_k \cdot \sqrt{\frac{\rho_k}{D \cdot (\rho_f - \rho_g) \cdot g}}$$
(14)

$$- y = \frac{f_{Wfy}}{(1-\alpha)^3} j_{fy}^* \cdot \sqrt{j_{fy}^{*2} + j_{fx}^{*2}} + \frac{j_{fy}^* f_{iy}}{\alpha(1-\alpha)^2} \cdot \sqrt{\left(\frac{j_{gx}^*}{\alpha} \sqrt{\frac{\rho_g}{\rho_f}} - \frac{j_{fx}^*}{1-\alpha} \frac{\rho_g}{\rho_f}\right)^2 + \frac{j_{fy}^{*2}}{(1-\alpha)^2} \left(\frac{\rho_g}{\rho_f}\right)^2} = 2$$
(15)

$$- X
\frac{f_{Wfy}}{(1-\alpha)^3} \dot{j}_{fx}^* \cdot \sqrt{j_{fy}^{*2} + j_{fx}^{*2}}
- \frac{\dot{j}_{gx}^{*2} \cdot f_{Wgx}}{\alpha^3} - f_{ix} \left[\frac{\dot{j}_{gx}^*}{\alpha^2 (1-\alpha)} - \frac{\dot{j}_{fx}^*}{\alpha (1-\alpha)^2} \sqrt{\frac{\rho_g}{\rho_f}} \right] \cdot \sqrt{\left(\frac{\dot{j}_{gx}^*}{\alpha} - \frac{\dot{j}_{fx}^*}{1-\alpha} \sqrt{\frac{\rho_g}{\rho_f}} \right)^2 + \frac{\dot{j}_{fy}^{*2}}{(1-\alpha)^2} \left(\frac{\rho_g}{\rho_f} \right)} = 0$$
(16)

$$\dot{m}_{f,tot} = \dot{m}_{fy} + \dot{m}_{fx} = \rho_f \upsilon_f D_1 W + \rho_f u_f D_2 H$$
(17)

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(17)
$$\sqrt{\frac{\rho_f}{D(\rho_f - \rho_g)g}}$$
 Wallis , (18)

$$j_{f,tot}^* = j_{f,y}^* + j_{f,x}^* \frac{H}{W}$$
(18)

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2) Annular flow

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, annular flow 가

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$$- y$$

$$\frac{f_{Wfy}}{(1-\alpha)^3} j_{fy}^* \cdot \sqrt{j_{fy}^{*2} + j_{fx}^{*2}} + \frac{j_{fy}^* f_{iy}}{\alpha (1-\alpha)^2} \cdot \sqrt{\left(\frac{j_{gx}^*}{\alpha} \sqrt{\frac{\rho_g}{\rho_f}} - \frac{j_{fx}^*}{1-\alpha} \frac{\rho_g}{\rho_f}\right)^2 + \frac{j_{fy}^{*2}}{(1-\alpha)^2} \left(\frac{\rho_g}{\rho_f}\right)^2} = 1$$

$$- X$$

$$\frac{f_{Wfy}}{(1-\alpha)^3} j_{fx}^* \cdot \sqrt{j_{fy}^{*2} + j_{fx}^{*2}}$$
(19)

$$-f_{ix}\left[\frac{j_{gx}^{*}}{\alpha^{2}(1-\alpha)}-\frac{j_{fx}^{*}}{\alpha(1-\alpha)^{2}}\sqrt{\frac{\rho_{g}}{\rho_{f}}}\right]\cdot\sqrt{\left(\frac{j_{gx}^{*}}{\alpha}-\frac{j_{fx}^{*}}{1-\alpha}\sqrt{\frac{\rho_{g}}{\rho_{f}}}\right)^{2}+\frac{j_{fy}^{*2}}{(1-\alpha)^{2}}\left(\frac{\rho_{g}}{\rho_{f}}\right)}=0$$

$$-$$

$$(20)$$

$$j_{f,tot}^{*} = j_{f,y}^{*} + j_{f,x}^{*} \frac{H}{W}$$
(21)

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(15)~(17) (19)~(21)
$$j_{f,tot}^*$$
 j_{gx}^*
 $\alpha, j_{fx}^*, j_{fy}^*, f_{ix}, f_{iy}, f_{wfx}, f_{wfy}, f_{wgx}, f_{wgy}$.

$$f_{wk} = \frac{0.046}{\text{Re}_k^{0.2}}$$
(22)
, Reynolds

$$\operatorname{Re}_{k} = \frac{\rho_{k} u_{k} D_{hk}}{\mu_{k}}, \qquad D_{hf} = \frac{4A_{fx}}{S_{fWx}}, \qquad D_{hg} = \frac{4A_{gx}}{S_{fgx} + S_{gWx}}$$
(23)

impinging	jet	

Wallis

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- Wallis [14]:
$$f_i = 0.005 [1+75(1-\alpha)]$$

가

- C.H.Newton et al.[15]:
$$f_i = 6.5 \cdot 10^{-4} \operatorname{Re}_g^{0.3}$$





6. DVI-4



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jet impingement . , 10 . 2 jet impingement7⊦ ,

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. , 10 impinging jet . onset of sweep-out [4]. 7

-1 , jet impingement -2 -3

11 1/7 DIVA . 가 impinging jet annular flow Newton Lee 3 . 가 . DVI-4 가 가 Wallis , Newton Lee 가 . Impinging jet Newton Lee 가 Wallis , 가 , Newton Lee

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10. DVI-2



12. 1//7 DIVA

: annular flow 가

DVI-4 7 12 . Imping jet

가 가 impinging jet 기

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, impinging jet

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j [*]	Wallis parameter	
'n		
Re	Reynolds	
S	Perimeter	
u	x	
v	У	
W	3	
α		
ρ		
τ		
f		
g		
i		
w		

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